

# Estimation of heat-transfer characteristics on a vertical annular circular fin of finned-tube heat exchangers in forced convection

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## Abstract

The finite difference method in conjunction with the least-squares scheme and experimental measured temperatures is applied to predict the average heat transfer coefficient  $\bar{h}$  and fin efficiency  $\eta_f$  on a vertical annular circular fin of finned-tube heat exchangers for various fin spacings in forced convection. The distribution of the heat transfer coefficient on the fin is assumed to be non-uniform, thus the whole annular circular fin is divided into several sub-fin regions in order to predict the  $\bar{h}$  and  $\eta_f$  values. These two predicted values can be obtained using the present inverse scheme in conjunction with the experimental measured temperatures. The results show that the effect of the fin spacing  $S$  on the  $\bar{h}$  value can be negligible when the  $S$  value exceeds about 0.018 m. The  $\bar{h}$  value increases with increasing the air speed  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$  ( $1550 \leq Re_d \leq 7760$ ) and increasing the fin spacing for  $0.005 \text{ m} \leq S \leq 0.018 \text{ m}$ . The present estimated results can be applied to obtain the new correlations of the Nusselt number and fin efficiency based on  $d_o$ ,  $S$  and  $Re_d$ . The average heat transfer coefficients obtained from this new correlation of the Nusselt number are in good agreement with those given by Hu and Jacobi [X. Hu, A.M. Jacobi, Local heat transfer behavior and its impact on a single-row, annularly finned tube heat exchanger, ASME J. Heat Transfer 115 (1993) 66–74].

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## 1. Introduction

Annular-finned tube heat exchangers are commonly used in industry. In designing such heat exchangers, it is necessary to note the interactions between the local heat transfer and flow distribution within the fins. The previous works about the effect of the fin spacing of annular-finned tube heat exchangers were limited to the experiments [1]. Thus the present study applies the hybrid inverse scheme in conjunction with experimental temperature data to estimate the heat-transfer characteristics of annular-finned tube heat exchangers in forced convection. The fin in heat exchangers is always applied to increase the heat flow per unit of basic surface. The analysis of a continuous plate fin pierced by a regularly spaced array of circular tubes in staggered and in-line arrays has many engineering appli-

cations [2]. In order to simplify the problem considered, the calculation of the standard fin efficiency usually assumed that the heat transfer coefficient was constant over the plate fin. It can be found from Refs. [3–6] that there exhibited very complex three-dimensional flow characteristics within a plate finned-tube heat exchanger. In particular, complex flow patterns caused by the horseshoe vortices at the corner junction are of interest to the study of local heat transfer enhancement mechanism. The flow accelerates around a heated horizontal annular-finned circular tube in a cross-flow and forms a low-velocity wake region behind the tube. The boundary layer over a heated horizontal tube starts to develop at the front of the tube and increases in thickness along the circumference of the tube. Thus the heat transfer coefficient in forced convection is highest on the upstream fin region and is lowest on the wake fin region, as shown in Ref. [7]. This causes local variations of the heat transfer coefficient on the fin. On the other hand, the heat transfer coefficient on the fin is non-uniform. As shown in Ref. [8],

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## Nomenclature

$A_f$	area of the annular circular fin, $m^2$	$r$	spatial coordinate
$A_j$	area of the $j$ th sub-fin region, $m^2$	$R_i$	inner radius of the annular circular fin or outer radius of the circular tube, m
$[A]$	global conduction matrix	$R_o$	outer radius of the annular circular fin, m
$D_H$	hydraulic diameter of heat exchanger, m	$Re$	Reynolds number defined in Eq. (23)
$d_o$	outer diameter of the circular tube, m	$Re_d$	Reynolds number defined in Eq. (24)
$[F]$	force matrix	$S$	fin spacing, m
$g$	acceleration of gravity, $m/s^2$	$T$	fin temperature, K
$h$	local heat transfer coefficient, $W/m^2 K$	$T_f$	film temperature, $(T_o + T_\infty)/2$
$\bar{h}$	unknown average heat transfer coefficient, $W/m^2 K$	$T_j$	measured temperature on the $j$ th sub-fin region, K
$\bar{h}^{iso}$	unknown average heat transfer coefficient under the isothermal condition, $W/m^2 K$	$T_o$	outer surface temperature of the circular tube, K
$h_f$	fin height, $R_o - R_i$	$T_\infty$	ambient temperature, K
$\bar{h}_j$	unknown average heat transfer coefficient on the $j$ th sub-fin region, $W/m^2 K$	$[T]$	global temperature matrix
$k_f$	thermal conductivity of the fin, $W/m K$	$V_{air}$	frontal air speed, m/s
$k_{air}$	thermal conductivity of the air, $W/m K$	$V_{max}$	maximum air speed, m/s
$\ell_r$	distance between two neighboring nodes in the $r$ -direction, $\ell_r = (R_o - R_i)/(N_r - 1)$	<i>Greek symbols</i>	
$\ell_\theta$	distance between two neighboring nodes in the $\theta$ -direction, $\ell_\theta = 2\pi/(N_\theta - 1)$	$\alpha$	thermal diffusivity of the air, $m^2/s$
$N$	number of sub-fin regions	$\beta$	volumetric thermal expansion coefficient, $1/K$
$Nu_d^{iso}$	Nusselt number defined in Eq. (22)	$\delta$	fin thickness, m
$N_r$	number of nodes in the $r$ -direction	$\eta_f$	fin efficiency
$N_\theta$	number of nodes in the $\theta$ -direction	$\nu$	kinematic viscosity of the air, $m^2/s$
$Q$	total heat transfer rate dissipated from the annular circular fin, W	$\theta$	spatial coordinate
$q_j$	heat transfer rate dissipated from the $j$ th sub-fin region, W	<i>Superscripts</i>	
		cal	calculated value
		mea	measured data

the measurement of the local heat transfer coefficient on the plate fin under steady-state heat transfer conditions was very difficult to perform, since the local fin temperature and local heat flux were required. Moreover, reliability is an important concept in engineering design, and the use of reliable components enables the designers to utilize more sophisticated techniques to improve the performance [9]. Thus the estimation of a more accurate heat transfer coefficient on the fin is an important task for the device of the high-performance heat exchangers.

Heat transfer coefficients encountered in forced convection are typically much higher than those encountered in natural convection because of the higher fluid velocities associated with forced convection. As a result, most of researchers tend to ignore natural convection in heat transfer analyses that involve forced convection. However, this error may be considerable at low velocities associated with forced convection [10].

It is known that the physical quantities of the test material can be predicted using the measured temperatures inside this test material. Such problems are called the inverse heat conduction problems that have become an

interesting subject recently. To date, various inverse methods in conjunction with the measured temperatures inside the test material have been developed for the analysis of the inverse heat conduction problems [11,12]. However, to the authors' knowledge, a few researchers performed the prediction of the local heat transfer coefficients on a vertical annular circular fin of finned-tube heat exchangers with regard for the effect of the fin spacing [3–7,13,15–17].

Sung et al. [3] applied the naphthalene sublimation technique to obtain the local mass transfer coefficient on a circular cylinder with transverse two annular fins in a cross flow. Hu and Jacobi [6] also applied the naphthalene sublimation technique to obtain the local convective heat and mass transfer coefficients on a circular cylinder with transverse annular fins in a cross flow over the Reynolds number ranging from 3300 to 12,000. Watel et al. [14] investigated the influence of the flow velocity and fin spacing on the forced convective heat transfer from a single annular-finned tube using the Particle Image Velocimetry and infrared thermography. They obtained a valuable correlation of the mean Nusselt number on the fin as a function of the dimensionless fin spacing, outer diameter of a circular tube

and Reynolds number for  $2550 \leq Re_d \leq 42,000$ . They also found that the reduction in fin spacing led to a decrease of heat transfer for a fixed Reynolds number. However, the difference of the average heat transfer coefficients on the downstream and upstream fin regions and fin efficiency was not shown in Ref. [14]. As stated by Hu and Jacobi [6], it was difficult to obtain fin efficiency measurements, because the temperature measurements within the fin or on its surface are inherently difficult to make without disrupting the heat transfer behavior. Mon and Gross [15] applied the three-dimensional numerical study to investigate the effect of the fin spacing on four-row annular-finned tube bundles in staggered and in-line arrangements. The heat transfer and fluid flow characteristics were predicted using the computational fluid dynamics commercial code of FLUENT. It can be found from Ref. [15] that the laminar flow between the fins and one-dimensional heat conduction equation were assumed. Sparrow and Samie [16] measured Nusselt numbers and pressure loss coefficients for one- and two-row arrays of annular-finned tubes using the experimental method. Matos et al. [17] applied a theoretical, numerical and experimental study to demonstrate that finned and non-finned circular and elliptic tubes heat exchangers in forced convection that can be optimized for maximum heat transfer under a fixed volume constraint. Jang [18] used the numerical and experimental studies to estimate the heat transfer and fluid flow characteristics of a four-row annular-finned tube heat exchanger in a staggered arrangement in order to develop the high efficiency air-cooled steam condenser for the power plant. Sometimes, it is maybe difficult to measure the temperature distributions on the fin of plate finned-tube heat exchangers using the infrared thermography and thermocouples for some practical heat transfer problems. Due to this reason, Chen and Hsu [13] applied the finite difference method in conjunction with the least-squares scheme and experimental measured temperatures to predict the natural-convection heat transfer coefficient and fin efficiency on a vertical annular circular fin of finned-tube heat exchangers in a small open box with regard for the effect of the fin spacing. It can be found that the predicted results of the average heat transfer coefficient given in Ref. [13] agreed with those obtained from the correlations recommended by current textbooks [1,19] under the assumption of the ideal isothermal fin. This implies that the present inverse scheme has good reliability. In order to validate the reliability and accuracy of the present inverse scheme in forced convection further, the present study performs the estimations of the average heat transfer coefficient and fin efficiency on a vertical annular circular fin of finned-tube heat exchangers in forced convection. The present estimated results of the average heat transfer coefficient under the ideal isothermal condition will compare with the estimated results given by Hu and Jacobi [6] and Watel et al. [14].

The inverse analysis of the present study is that the whole fin area is divided into several analysis sub-fin regions and then the fin temperatures at selected measure-

ment locations are measured using T-type thermocouples. Later, the finite difference method in conjunction with the measured temperatures and least-squares method is applied to predict the average heat transfer coefficients on these sub-fin regions. Furthermore, the average heat transfer coefficient  $\bar{h}$  and fin efficiency can be obtained for various air speeds and fin spacings under the given conditions of the ambient and tube temperatures. The computational procedures for the estimates of the heat transfer coefficient on each sub-fin region are performed repeatedly until the sum of the squares of the deviations between the calculated and measured temperatures becomes minimum.

## 2. Mathematical formulation

The schematic diagram of the present problem is shown in Fig. 1. Fig. 2 shows the physical model of the two-dimensional thin annular circular fin with measurement locations and sub-fin regions in forced convection.  $R_o$  and  $R_i$  denote the outer and inner radii of the annular circular fin, respectively.  $S$  and  $\delta$ , respectively, denote the fin spacing and fin thickness. The center of the circular tube is located at  $r = 0$ .  $T_o$  and  $T_\infty$ , respectively, denote the outer surface temperature of the circular tube and the ambient temperature. Due to the thin fin behavior, the temperature gradient in the  $z$ -direction (the fin thickness) is small and the fin temperature varies only in the  $r$ - and  $\theta$ -directions. The “insulated tip” assumption can be an adequate approximation provided that the actual heat transfer rate dissipated through the tip is much smaller than the total heat transfer rate drawn from the base wall [20]. It can be found from Refs. [7,13] that the heat transfer coefficient on the fin of finned-tube heat exchangers was non-uniform. Thus the heat transfer coefficient  $h(r, \theta)$  in the present study is also assumed to be non-uniform. The average heat trans-

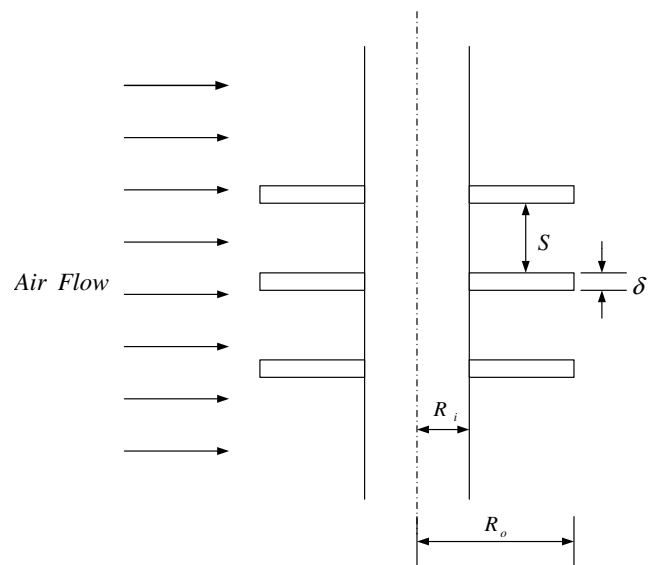


Fig. 1. Schematic diagram of one-tube annular-finned tube heat exchangers in forced convection.

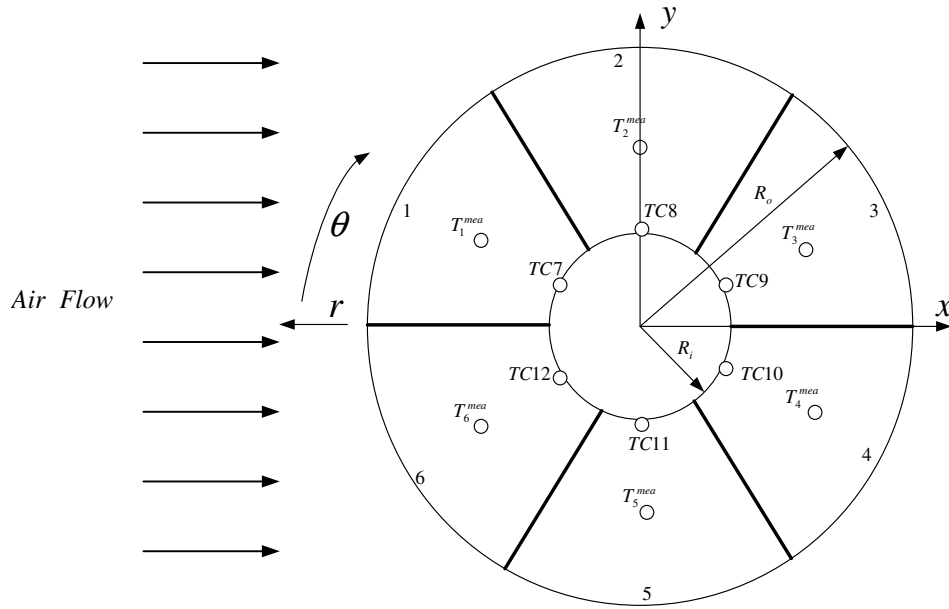


Fig. 2. Physical model of a two-dimensional annular circular fin with measurement locations and sub-fin regions in forced convection.

fer coefficient on each sub-fin region can be estimated provided that the fin temperatures at various measurement locations can be measured. Under the assumptions of the steady state and constant thermal properties, the two-dimensional heat conduction equation for the continuous the fin of a one-tube finned-tube heat exchanger can be expressed as

$$\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{1}{r^2} \frac{\partial^2 T}{\partial \theta^2} = \frac{2h(r, \theta)}{k_f \delta} (T - T_\infty) \text{ in } R_i < r < R_o, 0 < \theta \leq 2\pi \quad (1)$$

Its corresponding boundary conditions are

$$\frac{\partial T(r, 0)}{\partial \theta} = \frac{\partial T(r, 2\pi)}{\partial \theta} \quad (2)$$

$$T(r, 0) = T(r, 2\pi) \quad (3)$$

$$T(r, \theta) = T_o \text{ at } r = R_i \quad (4)$$

and

$$\frac{\partial T(r, \theta)}{\partial r} = 0 \text{ at } r = R_o \quad (5)$$

where  $T$  is the fin temperature.  $r$  and  $\theta$  are cylindrical coordinates.  $h(r, \theta)$  is the unknown heat transfer coefficient on the fin.  $k_f$  is the thermal conductivity of the fin.

### 3. Numerical analysis

It might be difficult to measure the temperature distributions on the annular circular fin using the infrared thermography and thermocouples for some practical heat transfer problems. Relatively, the distribution of the unknown heat transfer coefficient on a fin  $h(r, \theta)$  is not easy to be obtained. Under this circumstance, the annular circu-

lar fin considered can be divided into  $N$  sub-fin regions in the present inverse scheme and then the unknown heat transfer coefficient on each sub-fin region can be assumed to be constant. Thus the application of the finite difference method to Eq. (1) can produce the following difference equation on the  $k$ th sub-fin region as

$$\begin{aligned} & \frac{T_{i+1,j} - 2T_{i,j} + T_{i-1,j}}{\ell_r^2} + \frac{1}{R_i + (i-1)\ell_r} \frac{T_{i+1,j} - T_{i-1,j}}{2\ell_r} \\ & + \frac{1}{[R_i + (i-1)\ell_r]^2} \frac{T_{i,j+1} - 2T_{i,j} + T_{i,j-1}}{\ell_\theta^2} \\ & = \frac{2\bar{h}_k}{k_f \delta} (T_{i,j} - T_\infty) \end{aligned} \quad (6)$$

for  $i = 2, \dots, N_r,$   
 $j = (N_\theta - 1)(k - 1)/N + 2, (N_\theta - 1)(k - 1)/N + 3, \dots,$   
 $(N_\theta - 1)k/N$

where  $\ell_r$  and  $\ell_\theta$ , respectively, are the distances between two neighboring nodes in the  $r$ - and  $\theta$ -directions and are defined as  $\ell_r = (R_o - R_i)/(N_r - 1)$  and  $\ell_\theta = 2\pi/(N_\theta - 1)$ .  $N_r$  and  $N_\theta$  are the nodal numbers in the  $r$ - and  $\theta$ -directions, respectively.  $\bar{h}_k$  is denoted as the average heat transfer coefficient on the  $k$ th sub-fin region.

The application of the central difference approximation to the boundary condition (2) and then the substitution of the resulting equation into Eq. (6) can yield the difference equations as

$$\begin{aligned} & \frac{T_{i+1,1} - 2T_{i,1} + T_{i-1,1}}{\ell_r^2} + \frac{1}{R_i + (i-1)\ell_r} \frac{T_{i+1,1} - T_{i-1,1}}{2\ell_r} \\ & + \frac{1}{[R_i + (i-1)\ell_r]^2} \frac{T_{i,2} - 2T_{i,1} + T_{i,N_\theta-1}}{\ell_\theta^2} \\ & = \frac{2\bar{h}_m}{k_f \delta} (T_{i,1} - T_\infty) \text{ for } i = 2, 3, \dots, N_r \end{aligned} \quad (7)$$

and

$$\begin{aligned} & \frac{T_{i+1,N_\theta} - 2T_{i,N_\theta} + T_{i-1,N_\theta}}{\ell_r^2} + \frac{1}{R_i + (i-1)\ell_r} \frac{T_{i+1,N_\theta} - T_{i-1,N_\theta}}{2\ell_r} \\ & + \frac{1}{[R_i + (i-1)\ell_r]^2} \frac{T_{i,2} - 2T_{i,N_\theta} + T_{i,N_\theta-1}}{\ell_\theta^2} \\ & = \frac{2\bar{h}_m}{k_f \delta} (T_{i,N_\theta} - T_\infty) \quad \text{for } i = 2, 3, \dots, N_r \end{aligned} \quad (8)$$

where  $\bar{h}_m$  is defined as  $\bar{h}_m = (\bar{h}_1 + \bar{h}_N)/2$ .

The discretized form of Eq. (4) is given as

$$T_{1,j} = T_o \quad \text{for } j = 1, 2, \dots, N_\theta \quad (9)$$

The application of the central difference approximation to the boundary conditions (3) and (5) can yield their difference equations as

$$T_{i,1} = T_{i,N_\theta} \quad \text{for } i = 2, 3, \dots, N_r \quad (10)$$

and

$$T_{N_r-1,j} = T_{N_r+1,j} \quad \text{for } j = 1, 2, \dots, N_\theta \quad (11)$$

The difference equations for the nodes at the interface of two neighboring sub-fin regions, as shown in Fig. 3, can be expressed as

$$\begin{aligned} & \frac{T_{i+1,j} - 2T_{i,j} + T_{i-1,j}}{\ell_r^2} + \frac{1}{R_i + (i-1)\ell_r} \frac{T_{i+1,j} - T_{i-1,j}}{2\ell_r} \\ & + \frac{1}{[R_i + (i-1)\ell_r]^2} \frac{T_{i,j+1} - 2T_{i,j} + T_{i,j-1}}{\ell_\theta^2} \\ & = \frac{\bar{h}_k + \bar{h}_{k+1}}{k_f \delta} (T_{i,j} - T_\infty) \end{aligned} \quad (12)$$

for  $i = 2, \dots, N_r, j = (N_\theta - 1)k/N + 1, k = 1, 2, \dots, N - 1$

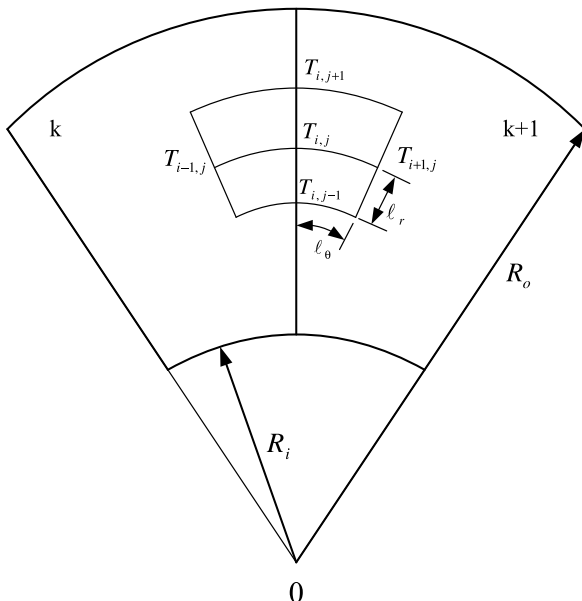


Fig. 3. Nodes at the interface of two-neighboring sub-fin areas.

Rearrangement of Eqs. (6)–(12) can yield the following matrix equation as

$$[A][T] = [F] \quad (13)$$

where  $[A]$  is a global conduction matrix.  $[T]$  is a matrix representing the nodal temperatures.  $[F]$  is a force matrix. The nodal temperatures can be obtained from Eq. (13) using the Gauss elimination algorithm.

Once the average heat transfer coefficient on each sub-fin region is obtained, the heat transfer rate dissipated from the  $j$ th sub-fin region  $q_j$  and average heat transfer coefficient on the whole fin  $\bar{h}$  can be determined using the following expressions. The heat transfer rate dissipated from the  $j$ th sub-fin region  $q_j$  is defined as

$$q_j = 2\bar{h}_j \int_{A_j} (T - T_\infty) dA \quad \text{for } j = 1, 2, \dots, N \quad (14)$$

The average heat transfer coefficient on the whole fin  $\bar{h}$  can be expressed as

$$\bar{h} = \sum_{j=1}^N \bar{h}_j A_j / A_f \quad (15)$$

where  $A_f$  is the area of the annular circular fin.  $A_j$  is the area of the  $j$ th sub-fin region.

The fin efficiency  $\eta_f$  is defined as the ratio of the actual heat transfer rate from the whole fin to the dissipated heat transfer rate from the fin maintained at the tube temperature  $T_o$ . Thus the fin efficiency  $\eta_f$  can be expressed as

$$\eta_f = \frac{\sum_{j=1}^N q_j}{2A_f (T_o - T_\infty) \bar{h}} \quad (16)$$

The total heat transfer rate dissipated from the whole fin to the ambient  $Q$  can be written as

$$Q = \sum_{j=1}^N q_j \quad (17)$$

In order to estimate the unknown heat transfer coefficient on the  $j$ th sub-fin region  $\bar{h}_j$ , the additional information of the steady-state measured temperatures is required. The more a number of the sub-fin regions are, the more accurate the estimation of the unknown average heat transfer coefficient on the whole fin can be. Relatively, a more computational time can be required. In the present study, T-type thermocouples are used to record the temperature information at selected measurement locations. The measured temperature taken from the  $j$ th thermocouple is denoted as  $T_j^{mea}, j = 1, \dots, N$ , as shown in Tables 1–3.

The least-squares minimization technique is applied to minimize the sum of the squares of the deviations between the calculated and measured temperatures at selected measurement locations. In the present study, the unknown average heat transfer coefficients on each sub-fin region  $\bar{h}_i$  can be expressed as

$$\bar{h}_i = C_i \quad \text{for } i = 1, 2, \dots, N \quad (18)$$

Table 1  
Temperature measurements and the present estimates for  $V_{\text{air}} = 1$  m/s and various  $S$  values

	$S = 0.005$ m	$S = 0.01$ m	$S = 0.015$ m	$S \rightarrow \infty$
	$T_0 = 330.63$ K $T_\infty = 298.03$ K	$T_0 = 328.82$ K $T_\infty = 298.20$ K	$T_0 = 330.34$ K $T_\infty = 299.74$ K	$T_0 = 335.03$ K $T_\infty = 300.23$ K
$T_j^{\text{mea}}$ (K)	$T_1^{\text{mea}} = 305.67$ $T_2^{\text{mea}} = 305.83$ $T_3^{\text{mea}} = 314.30$ $T_4^{\text{mea}} = 312.93$ $T_5^{\text{mea}} = 305.91$ $T_6^{\text{mea}} = 303.70$	$T_1^{\text{mea}} = 304.71$ $T_2^{\text{mea}} = 305.36$ $T_3^{\text{mea}} = 309.22$ $T_4^{\text{mea}} = 309.28$ $T_5^{\text{mea}} = 305.39$ $T_6^{\text{mea}} = 303.17$	$T_1^{\text{mea}} = 305.51$ $T_2^{\text{mea}} = 305.67$ $T_3^{\text{mea}} = 313.25$ $T_4^{\text{mea}} = 311.31$ $T_5^{\text{mea}} = 307.67$ $T_6^{\text{mea}} = 304.75$	$T_1^{\text{mea}} = 306.67$ $T_2^{\text{mea}} = 307.06$ $T_3^{\text{mea}} = 312.75$ $T_4^{\text{mea}} = 312.75$ $T_5^{\text{mea}} = 309.22$ $T_6^{\text{mea}} = 306.16$
$\bar{h}_j$ (W/m <sup>2</sup> K)	$\bar{h}_1 = 24.72$ $\bar{h}_2 = 35.86$ $\bar{h}_3 = 3.20$ $\bar{h}_4 = 7.49$ $\bar{h}_5 = 31.02$ $\bar{h}_6 = 45.46$	$\bar{h}_1 = 29.65$ $\bar{h}_2 = 31.97$ $\bar{h}_3 = 12.51$ $\bar{h}_4 = 12.51$ $\bar{h}_5 = 29.40$ $\bar{h}_6 = 48.96$	$\bar{h}_1 = 34.35$ $\bar{h}_2 = 45.10$ $\bar{h}_3 = 4.06$ $\bar{h}_4 = 13.98$ $\bar{h}_5 = 24.56$ $\bar{h}_6 = 48.50$	$\bar{h}_1 = 36.27$ $\bar{h}_2 = 41.00$ $\bar{h}_3 = 11.20$ $\bar{h}_4 = 13.62$ $\bar{h}_5 = 24.74$ $\bar{h}_6 = 45.60$
$q_j$ (W)	$q_1 = 0.72$ $q_2 = 1.11$ $q_3 = 0.14$ $q_4 = 0.32$ $q_5 = 0.95$ $q_6 = 1.11$	$q_1 = 0.77$ $q_2 = 0.90$ $q_3 = 0.43$ $q_4 = 0.43$ $q_5 = 0.82$ $q_6 = 1.09$	$q_1 = 0.85$ $q_2 = 1.19$ $q_3 = 0.16$ $q_4 = 0.51$ $q_5 = 0.72$ $q_6 = 1.08$	$q_1 = 1.01$ $q_2 = 1.22$ $q_3 = 0.44$ $q_4 = 0.54$ $q_5 = 0.82$ $q_6 = 1.17$
$\bar{h}$ (W/(m <sup>2</sup> K))	24.63	27.50	28.42	28.74
$\bar{h}^{\text{iso}}$ W/(m <sup>2</sup> K)	Present Eq. (20) Present Eq. (25) Ref. [14]	18.73 18.49 18.49	20.35 19.37 21.04	20.69 19.55 21.98
$Q$ (W)	4.35	4.44	4.51	5.20
$\eta_f$	38%	37%	36%	35%

Table 2  
Temperature measurements and the present estimates for  $V_{\text{air}} = 3$  m/s and various  $S$  values

	$S = 0.005$ m	$S = 0.01$ m	$S = 0.015$ m	$S \rightarrow \infty$
	$T_0 = 331.70$ K $T_\infty = 298.15$ K	$T_0 = 331.00$ K $T_\infty = 299.00$ K	$T_0 = 330.80$ K $T_\infty = 299.90$ K	$T_0 = 331.60$ K $T_\infty = 300.15$ K
$T_j^{\text{mea}}$ (K)	$T_1^{\text{mea}} = 300.31$ $T_2^{\text{mea}} = 303.73$ $T_3^{\text{mea}} = 310.39$ $T_4^{\text{mea}} = 308.82$ $T_5^{\text{mea}} = 302.16$ $T_6^{\text{mea}} = 301.31$	$T_1^{\text{mea}} = 302.46$ $T_2^{\text{mea}} = 302.75$ $T_3^{\text{mea}} = 306.36$ $T_4^{\text{mea}} = 306.92$ $T_5^{\text{mea}} = 302.79$ $T_6^{\text{mea}} = 301.21$	$T_1^{\text{mea}} = 302.49$ $T_2^{\text{mea}} = 302.95$ $T_3^{\text{mea}} = 307.21$ $T_4^{\text{mea}} = 306.95$ $T_5^{\text{mea}} = 303.41$ $T_6^{\text{mea}} = 302.46$	$T_1^{\text{mea}} = 302.95$ $T_2^{\text{mea}} = 303.08$ $T_3^{\text{mea}} = 306.79$ $T_4^{\text{mea}} = 307.05$ $T_5^{\text{mea}} = 303.80$ $T_6^{\text{mea}} = 302.79$
$\bar{h}_j$ (W/m <sup>2</sup> K)	$\bar{h}_1 = 116.86$ $\bar{h}_2 = 45.01$ $\bar{h}_3 = 9.38$ $\bar{h}_4 = 14.13$ $\bar{h}_5 = 69.25$ $\bar{h}_6 = 73.28$	$\bar{h}_1 = 62.67$ $\bar{h}_2 = 67.67$ $\bar{h}_3 = 25.12$ $\bar{h}_4 = 20.89$ $\bar{h}_5 = 64.38$ $\bar{h}_6 = 107.52$	$\bar{h}_1 = 84.40$ $\bar{h}_2 = 80.43$ $\bar{h}_3 = 21.83$ $\bar{h}_4 = 25.12$ $\bar{h}_5 = 67.88$ $\bar{h}_6 = 87.12$	$\bar{h}_1 = 78.63$ $\bar{h}_2 = 84.43$ $\bar{h}_3 = 27.02$ $\bar{h}_4 = 26.16$ $\bar{h}_5 = 65.85$ $\bar{h}_6 = 86.73$
$q_j$ (W)	$q_1 = 2.24$ $q_2 = 1.17$ $q_3 = 0.35$ $q_4 = 0.49$ $q_5 = 1.62$ $q_6 = 1.39$	$q_1 = 1.29$ $q_2 = 1.48$ $q_3 = 0.71$ $q_4 = 0.60$ $q_5 = 1.40$ $q_6 = 1.82$	$q_1 = 1.56$ $q_2 = 1.61$ $q_3 = 0.59$ $q_4 = 0.68$ $q_5 = 1.41$ $q_6 = 1.48$	$q_1 = 1.50$ $q_2 = 1.69$ $q_3 = 0.71$ $q_4 = 0.70$ $q_5 = 1.40$ $q_6 = 1.51$
$\bar{h}$ (W/(m <sup>2</sup> K))	54.65	58.04	61.13	61.47
$\bar{h}^{\text{iso}}$ (W/(m <sup>2</sup> K))	Present Eq. (20) Present Eq. (25) Ref. [14]	30.37 30.96 35.43	32.02 31.67 39.38	33.29 31.78 40.90
$Q$ (W)	7.26	7.30	7.33	7.51
$\eta_f$	28%	28%	27%	27%

Table 3  
Temperature measurements and the present estimates for  $V_{\text{air}} = 5$  m/s and various  $S$  values

	$S = 0.005$ m	$S = 0.01$ m	$S = 0.015$ m	$S \rightarrow \infty$
	$T_0 = 330.13$ K $T_\infty = 298.90$ K	$T_0 = 331.20$ K $T_\infty = 299.25$ K	$T_0 = 329.30$ K $T_\infty = 299.53$ K	$T_0 = 331.50$ K $T_\infty = 299.88$ K
$T_j^{\text{mea}}$ (K)	$T_1^{\text{mea}} = 300.98$ $T_2^{\text{mea}} = 301.37$ $T_3^{\text{mea}} = 306.43$ $T_4^{\text{mea}} = 308.75$ $T_5^{\text{mea}} = 300.98$ $T_6^{\text{mea}} = 300.86$	$T_1^{\text{mea}} = 301.80$ $T_2^{\text{mea}} = 301.97$ $T_3^{\text{mea}} = 305.11$ $T_4^{\text{mea}} = 305.41$ $T_5^{\text{mea}} = 301.67$ $T_6^{\text{mea}} = 300.82$	$T_1^{\text{mea}} = 301.18$ $T_2^{\text{mea}} = 301.64$ $T_3^{\text{mea}} = 304.43$ $T_4^{\text{mea}} = 305.25$ $T_5^{\text{mea}} = 301.48$ $T_6^{\text{mea}} = 301.18$	$T_1^{\text{mea}} = 301.44$ $T_2^{\text{mea}} = 302.39$ $T_3^{\text{mea}} = 304.13$ $T_4^{\text{mea}} = 304.92$ $T_5^{\text{mea}} = 302.13$ $T_6^{\text{mea}} = 301.61$
$\bar{h}_j$ (W/(m <sup>2</sup> K))	$\bar{h}_1 = 102.50$ $\bar{h}_2 = 98.83$ $\bar{h}_3 = 24.03$ $\bar{h}_4 = 7.68$ $\bar{h}_5 = 121.35$ $\bar{h}_6 = 107.11$	$\bar{h}_1 = 83.83$ $\bar{h}_2 = 90.25$ $\bar{h}_3 = 33.43$ $\bar{h}_4 = 29.36$ $\bar{h}_5 = 98.33$ $\bar{h}_6 = 137.38$	$\bar{h}_1 = 120.11$ $\bar{h}_2 = 103.57$ $\bar{h}_3 = 39.21$ $\bar{h}_4 = 27.64$ $\bar{h}_5 = 114.14$ $\bar{h}_6 = 119.16$	$\bar{h}_1 = 134.30$ $\bar{h}_2 = 89.38$ $\bar{h}_3 = 51.64$ $\bar{h}_4 = 37.36$ $\bar{h}_5 = 103.20$ $\bar{h}_6 = 120.46$
$q_j$ (W)	$q_1 = 1.79$ $q_2 = 1.87$ $q_3 = 0.68$ $q_4 = 0.24$ $q_5 = 2.24$ $q_6 = 1.67$	$q_1 = 1.56$ $q_2 = 1.77$ $q_3 = 0.84$ $q_4 = 0.75$ $q_5 = 1.86$ $q_6 = 2.10$	$q_1 = 1.91$ $q_2 = 1.79$ $q_3 = 0.88$ $q_4 = 0.65$ $q_5 = 1.95$ $q_6 = 1.72$	$q_1 = 2.22$ $q_2 = 1.67$ $q_3 = 1.14$ $q_4 = 0.87$ $q_5 = 1.89$ $q_6 = 1.84$
$\bar{h}$ (W/(m <sup>2</sup> K))	76.92	78.76	87.30	89.39
$\bar{h}^{\text{iso}}$ (W/(m <sup>2</sup> K))	Present Eq. (20) Present Eq. (25) Ref. [14]	38.15 39.27 47.20	39.01 39.79 52.67	41.96 39.83 54.58
$Q$ (W)	8.49	8.88	8.90	9.63
$\eta_r$	25%	25%	24%	24%

The error in the estimates  $E(C_1, C_2, \dots, C_N)$  will be minimized in order to obtain the required estimates.  $E(C_1, C_2, \dots, C_N)$  is defined as

$$E(C_1, C_2, \dots, C_N) = \sum_{j=1}^N [T_j^{\text{cal}} - T_j^{\text{mea}}]^2 \quad (19)$$

where  $T_j^{\text{cal}}$  denotes the calculated temperature taken from the  $j$ th thermocouple location and is obtained from Eq. (13).

The estimated values of  $C_j$ ,  $j = 1, 2, \dots, N$ , are determined until the value of  $E(C_1, C_2, \dots, C_N)$  is minimum. The detailed computational procedures for estimating the  $C_j$  values,  $j = 1, 2, \dots, N$ , can be found in Refs. [7,13]. In order to avoid repetition, they are not shown in this paper. The computational procedures of the present study are repeated until the values of  $\left| \frac{T_j^{\text{mea}} - T_j^{\text{cal}}}{T_j^{\text{mea}}} \right|$ ,  $j = 1, 2, \dots, N$ , are less than  $10^{-6}$ .

#### 4. Experimental apparatus

An experimental configuration of the small wind tunnel used for the present problem is the same as Fig. 1 of Ref. [7]. The experimental procedures and manners have also been shown in Ref. [7]. In order to avoid repetition, they are not shown in this study. Fig. 4 shows the experimental configuration of the test annular fin vertically mounted on

a circular tube in forced convection. The test annular circular fin with 27 mm in inner diameter, 99 mm in outer diameter and 1 mm in thickness are made of AISI 304 stainless material. It can be found from Ref. [21] that the thermal conductivity of AISI 304 stainless material is 14.9 W/m K. An anemometer installed at 300 mm in front of the airflow entering the test specimen is used to measure the frontal air velocity. The limit of its error is  $\pm 0.4\%$  for the velocity ranging from 0.4 m/s to 30.0 m/s. The ambient and test fin temperatures are measured using T-type thermocouples. The limit of error of the T-type thermocouple is  $\pm 0.4\%$  for  $0^\circ\text{C} \leq T \leq 350^\circ\text{C}$ . Six thermocouples, TC7, TC8, TC9, TC10, TC11 and TC12, placed in the gap between the fin and the circular tube are fixed at  $\pi/6$ ,  $\pi/2$ ,  $5\pi/6$ ,  $7\pi/6$ ,  $3\pi/2$  and  $11\pi/6$  in angle, as shown in Fig. 2. In order to reduce the heat loss between the fin and the circular tube, their gap is filled with the cyanoacrylate (Satlon, D-3). The average of these six measured temperatures is taken as the fin base temperature and is also assumed to be the outer surface temperature of the circular tube  $T_0$  in the present study. Three thermocouples penetrated the central line of the top surface and two lateral surfaces are positioned at 100 mm away from the test fin in order to measure the ambient temperature  $T_\infty$ . The average of these three measured temperatures is taken as the ambient temperature  $T_\infty$ . In order to estimate the average heat transfer coefficient on the fin, the annular circular fin

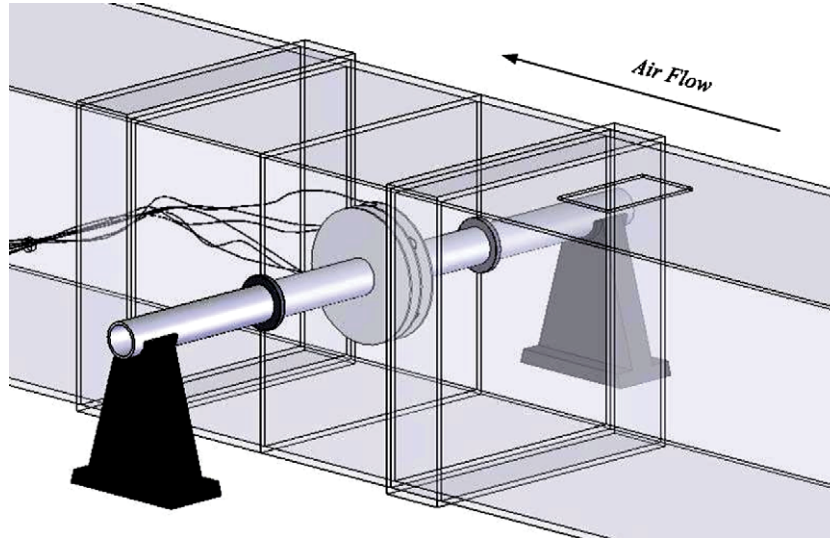


Fig. 4. Experimental configuration of the test annular fin vertically mounted on a circular tube in forced convection.

is divided into six regions in the present study, i.e.,  $N = 6$ . Thus six thermocouples are fixed at 33.5 mm in radius from the center of the circular tube and at  $\pi/6$ ,  $\pi/2$ ,  $5\pi/6$ ,  $7\pi/6$ ,  $3\pi/2$  and  $11\pi/6$  in angle, as shown in Fig. 2.

## 5. Results and discussion

The ratio of the surface area of the fin tip to the total fin surface area can be written as  $\frac{R_o \delta}{(R_o^2 - R_i^2) + R_o \delta}$ . The “insulated tip” assumption in the present study will be reasonable provided that this ratio is very small. For simplicity, the average heat transfer coefficient on the tip surface can be assumed to be the same as that on the lateral surfaces of the fin. Based on experiment data of the present study, the surface area of the fin tip is only 2.14% of the total fin surface area. This implies that the assumption of Eq. (5) should be reasonable.

All the computations are performed with  $N_r = 10$  and  $N_\theta = 49$ . The measured temperatures,  $T_1^{\text{mea}}(33.5, \pi/6)$ ,  $T_2^{\text{mea}}(33.5, \pi/2)$ ,  $T_3^{\text{mea}}(33.5, 5\pi/6)$ ,  $T_4^{\text{mea}}(33.5, 7\pi/6)$ ,  $T_5^{\text{mea}}(33.5, 3\pi/2)$  and  $T_6^{\text{mea}}(33.5, 11\pi/6)$ , are respectively denoted as  $T_1^{\text{mea}}$ ,  $T_2^{\text{mea}}$ ,  $T_3^{\text{mea}}$ ,  $T_4^{\text{mea}}$ ,  $T_5^{\text{mea}}$  and  $T_6^{\text{mea}}$ . They are shown in Tables 1–3 for  $V_{\text{air}} = 1$  m/s, 3 m/s and 5 m/s.  $T_1^{\text{mea}} \neq T_6^{\text{mea}}$ ,  $T_2^{\text{mea}} \neq T_5^{\text{mea}}$  and  $T_3^{\text{mea}} \neq T_4^{\text{mea}}$  can be observed from Tables 1–3. This result shows further that the symmetric assumptions of the flow and thermal fields are not always very reasonable for the present problem. The similar phenomenon can also be found from Ref. [7]. Tables 1–3 also show the effect of the fin spacing  $S$  on the average heat transfer coefficient on the  $j$ th sub-fin region  $\bar{h}_j$ , heat transfer rate on the  $j$ th sub-fin region  $q_j$ , total heat transfer rate on the whole fin  $Q$ , average heat transfer coefficient  $\bar{h}$  and fin efficiency  $\eta_f$ .

Due to the blockade of the tube, the airflow coming into the region between two parallel fins accelerates around a heated horizontal tube and forms a low-velocity wake

region behind the tube. The boundary layer starts to develop at the front of the heated tube and increases in thickness along the circumference of this tube. Thus there exists a complex flow pattern within an annular-finned tube heat exchanger in forced convection. It can be found from Tables 1–3 that the measured fin temperatures on the downstream sub-fin regions 3 and 4 are higher than those on the other sub-fin regions for various  $V_{\text{air}}$  and  $S$  values. In addition, the average heat transfer coefficients on the downstream sub-fin regions 3 and 4 behind the tube are lower than those on the other sub-fin regions because these regions belong to the low-performance wake regions. Moreover, due to the formation of recirculating flow, the average heat transfer coefficients on the sub-fin regions 2 and 5,  $\bar{h}_2$  and  $\bar{h}_5$ , can be larger than those on the downstream sub-fin regions 3 and 4,  $\bar{h}_3$  and  $\bar{h}_4$ . It can be observed Tables 1–3 that  $\bar{h}_2$  and  $\bar{h}_5$  increase with increasing  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$ . The ratio of the  $\bar{h}_2$  value for  $V_{\text{air}} = 3$  m/s and 5 m/s to that for  $V_{\text{air}} = 1$  m/s, respectively, is about 1.3 times and 2.8 times as  $S = 0.005$  m. This implies that the horseshoe vortex effect can become more obvious with increasing the air velocity. The similar result can be found from Refs. [6,15]. The ratio of the average heat transfer coefficients on the front fin regions to those on the wake fin regions,  $(\bar{h}_1 + \bar{h}_6)/(\bar{h}_3 + \bar{h}_4)$ , respectively is about 6 times, 7 times and 6.5 times for  $V_{\text{air}} = 1$  m/s, 3 m/s and 5 m/s as  $S = 0.005$  m. This result shows that the heat transfer coefficient is much higher on the entrance fin region than on the wake fin region. The ratio of the heat transfer rates on the front fin regions to those on the wake fin regions,  $(q_1 + q_6)/(q_3 + q_4)$ , respectively is about 3.98 times, 4.27 times and 3.9 times for  $V_{\text{air}} = 1$  m/s, 3 m/s and 5 m/s as  $S = 0.005$  m. The results in Tables 1–3 also show that the heat transfer rate on the wake fin regions of the present finned-tube heat exchanger from  $V_{\text{air}} = 1$  m/s to 5 m/s is responsible for 10.6–11.7% of the total heat transfer rate on the whole fin as  $S = 0.005$  m



and for 14.7–20.9% as  $S \geq 0.01$  m. The average heat transfer coefficient on each sub-fin region can be sensitive to the measured fin temperatures. However, the average heat transfer coefficient  $\bar{h}$  seems to be not very sensitive to the measured fin temperatures for the present inverse scheme.

As stated by Hu and Jacobi [6], it was difficult to obtain fin efficiency measurements, because thermocouples on the fin surface can significantly alter the flow and heat transfer over the surface. This implies that the measured fin temperatures were inherently difficult to make without disrupting the heat transfer behavior. It can be observed from Tables 1–3 that the average heat transfer coefficient  $\bar{h}$  increases with increasing the air speed  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$  ( $1550 \leq Re_d \leq 7760$ ) and increasing the fin spacing  $S$  for  $0.005 \text{ m} \leq S \leq 0.015 \text{ m}$ . The  $\bar{h}$  value in the range of  $V_{\text{air}} = 1 \text{ m/s}$  to  $5 \text{ m/s}$  increases from  $24.63 \text{ W/m}^2 \text{ K}$  to  $76.92 \text{ W/m}^2 \text{ K}$  for  $S = 0.005 \text{ m}$  and from  $28.42 \text{ W/m}^2 \text{ K}$  to  $87.30 \text{ W/m}^2 \text{ K}$  for  $S = 0.015 \text{ m}$ . The fin efficiency  $\eta_f$  decreases with increasing  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$  and seems to be not very sensitive to the fin spacing. The  $\eta_f$  value in the range of  $V_{\text{air}} = 1 \text{ m/s}$  to  $5 \text{ m/s}$  decreases from 0.38 to 0.25 for  $S = 0.005 \text{ m}$  and from 0.36 to 0.24 for  $S = 0.015 \text{ m}$ . It can be observed that the present estimates of the  $\eta_f$  value are higher than those obtained from a vertical square fin [7]. Fig. 5 shows variation of the  $\bar{h}$  value with  $S/h_f$  for various  $Re_d$  values, where  $h_f$  denotes the fin height and is equal to 36 mm in the present study. It can be found from Fig. 5 that the  $\bar{h}$  value approaches the asymptotical value obtained from a single annular circular fin as  $S \rightarrow \infty$ . The effect of the fin spacing  $S$  on the  $\bar{h}$  value can be negligible when the  $S/h_f$  value exceeds about 0.5. The similar phenomenon can also be observed from Ref. [7].

Based on Eqs. (14) and (17), the present estimate of the average heat transfer coefficient under the ideal isothermal

condition  $\bar{h}^{\text{iso}}$  can be determined from the following expression:

$$Q = \sum_{j=1}^N 2\bar{h}_j \int_{A_j} (T - T_\infty) dA = 2\bar{h}^{\text{iso}} A_f (T_o - T_\infty) \quad (20)$$

In order to validate the reliability of the present inverse scheme, Tables 1–4, respectively, show comparisons between the present estimates of the  $\bar{h}^{\text{iso}}$  value and those given by Hu and Jacobi [6] and Watel et al. [14].

It can be found from Ref. [14] that the correlated relationship of  $Nu_d^{\text{iso}}$  and  $Re_d$  for  $2550 \leq Re_d \leq 42000$  and various fin spacings was expressed as

$$Nu_d^{\text{iso}} = 0.446 \left[ \left( \frac{\delta}{S} + 1 \right) \left( 1 - \frac{K^*}{(S/d_o)^b (Re_d)^{0.07}} \right) \right]^{0.55} Re_d^{0.55} \quad (21)$$

where all the properties are evaluated at the film temperature  $T_f$ .  $Re_d$  and  $Nu_d^{\text{iso}}$  are respectively denoted as the Reynolds number based on the outer diameter of the circular tube and the Nusselt number based on the temperature difference of  $T_o$  and  $T_\infty$ .  $Nu_d^{\text{iso}}$  and  $Re_d$  are defined as

$$Nu_d^{\text{iso}} = \frac{\bar{h}_d^{\text{iso}} d_o}{k_{\text{air}}} \quad (22)$$

and

$$Re_d = \frac{V_{\text{air}} d_o}{\nu} \quad (23)$$

where  $k_{\text{air}}$  is the thermal conductivity of the air  $\nu$  is the kinematic viscosity.

The parameters  $b$  and  $K^*$  are given as

$$b = 0.27, \quad K^* = 0.62 \quad \text{for } 0.034 \leq S/d_o \leq 0.14$$

and

$$b = 0.55, \quad K^* = 0.36 \quad \text{for } 0.14 \leq S/d_o$$

In order to compare with the estimated results given by Hu and Jacobi [6], the Reynolds number based on the hydraulic diameter of the heat exchanger  $Re$  is introduced as

$$Re_d = \frac{V_{\text{max}} D_H}{\nu} \quad (24)$$

where  $V_{\text{max}}$  is the maximum flow speed at the minimum flow area.  $D_H$  is the hydraulic diameter of the heat exchanger.

In order to match the present estimates of  $Nu_d^{\text{iso}}$  and  $\eta_f$  for various  $Re_d$ ,  $S$  and  $d_o$  values, the modification of Eq. (21) in conjunction with the present estimates and the predicted results given by Hu and Jacobi [6] can yield the correlation of  $Nu_d^{\text{iso}}$ ,  $Re_d$ ,  $d_o$  and  $S$  as

$$Nu_d^{\text{iso}} = 16.5185 d_o \left[ \left( 2.54 \frac{\delta}{S} + 0.6925 \right) Re_d \times \left( 1 - \frac{K^*}{(S/d_o)^b (Re_d)^{0.07}} \right) \right]^{0.55} \left( \frac{1}{V_{\text{air}}} \right)^{0.123} \quad (25)$$

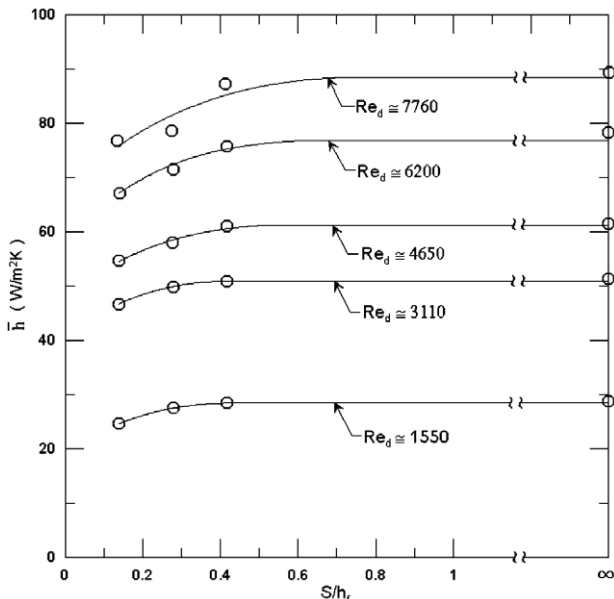


Fig. 5. Variation of the  $\bar{h}$  value with  $S/h_f$  for various  $Re_d$  values.

Table 4

Comparison of the  $\bar{h}^{\text{iso}}$  value between the present estimates and previous results for various  $V_{\text{air}}$  or  $Re$  values

			$\bar{h}^{\text{iso}}\text{W}/(\text{m}^2\text{K})$				
			$S = 0.005 \text{ m}$	$S = 0.0061 \text{ m}$	$S = 0.01 \text{ m}$	$S = 0.015 \text{ mm}$	$S \rightarrow \infty$
$V_{\text{air}} = 2 \text{ m/s}$	Present	Eq. (20)	26.93	–	29.36	29.53	29.81
		Eq. (25)	25.64	–	26.42	26.56	28.27
	Ref. [14]		27.92	–	31.82	32.53	37.68
$V_{\text{air}} = 4 \text{ m/s}$	Present	Eq. (20)	34.45	–	36.76	38.31	39.40
		Eq. (25)	35.40	–	36.00	36.07	38.03
	Ref. [14]		41.98	–	46.37	48.10	55.18
$Re = 3320$	Present	Eq. (25)	–	33.91	–	–	–
	Ref. [6]		–	31.6	–	–	–
$Re = 5980$	Present	Eq. (25)	–	44.7	–	–	–
	Ref. [6]		–	45.5	–	–	–
$Re = 6880$	Present	Eq. (25)	–	47.70	–	–	–
	Ref. [6]		–	49.5	–	–	–
$Re = 7580$	Present	Eq. (25)	–	49.94	–	–	–
	Ref. [6]		–	50.5	–	–	–
$Re = 8440$	Present	Eq. (25)	–	52.5	–	–	–
	Ref. [6]		–	55.8	–	–	–

The correlation of  $\eta_f$ ,  $Re_d$ ,  $d_o$  and  $S$  for the present estimates can be expressed as

$$\eta_f = \left(0.208 \frac{\delta}{S} + 1\right) [0.40783 - 4.17947 \times 10^{-5} Re_d + 2.598 \times 10^{-9} (Re_d)^2] \left(1 + \frac{25.61}{(S/d_o)^{0.55} Re_d}\right) \quad (26)$$

It can be found from Ref. [16] that the correlated relationship of  $Nu_d^{\text{iso}}$  and  $Re_m$  for  $S_T/(2R_o) \leq 1.52$  and  $(2R_o)/d_o = 1.8$  was expressed as

$$Nu_d^{\text{iso}} = 0.0529 Re_m^{0.704} \quad (27)$$

where all the properties are also evaluated at the film temperature  $T_f$ . The parameter  $S_T$  denotes the transverse pitch between two neighboring tube centers.  $Re_m$  is the Reynolds number and is defined as

$$Re_m = \frac{V_{\text{max}} d_o}{\nu} \quad (28)$$

In Eq. (28),  $V_{\text{max}}$  is the maximum flow speed at the minimum flow area.

Comparisons of  $\bar{h}^{\text{iso}}$  between the present estimates obtained from Eqs. (20) and (25) and those given by Watel et al. [14] are shown in Tables 1–4 for various  $S$  values and  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$ . It is found from Table 1 that the present estimates of the  $\bar{h}^{\text{iso}}$  value are in good agreement with those given by Watel et al. [14] for  $0.005 \text{ m} \leq S \leq 0.015 \text{ m}$  and  $V_{\text{air}} = 1 \text{ m/s}$ . However, inspection of Tables 1–4 reveals that the present estimates of the  $\bar{h}^{\text{iso}}$  value for  $V_{\text{air}} = 1 \text{ m/s}$  and  $S \rightarrow \infty$  and for  $V_{\text{air}} \geq 2 \text{ m/s}$  are lower than those given by Watel et al. [14]. This deviation seems to become large with increasing  $V_{\text{air}}$  or  $S$ . It may result from measurement locations and some experimental uncertainties such as personal fluctuations, circumstance of

the experiment and experimental measurements, etc. It is found from Ref. [14] that the measurement of the flow velocity and a thermocouple, respectively, was located at the wind tunnel section 280 mm and 80 mm upstream the fins. Obviously, these two measurement locations are different from the present measurement locations.

Comparisons of the  $\bar{h}^{\text{iso}}$  value between the present estimates obtained from Eq. (25) and those given by Hu and Jacobi [6] are also shown in Table 4 for  $S = 0.0061 \text{ m}$ ,  $d_o = 0.0381 \text{ m}$ ,  $\delta = 0.00102 \text{ m}$ ,  $T_o = 80.2 \text{ }^\circ\text{C}$ ,  $T_\infty = 27 \text{ }^\circ\text{C}$  and various  $Re$  values. Maybe, due to the inattention of the authors, their estimated results for  $Re = 5980$  and  $6880$  can be in error, as shown in Table 2 of Ref. [6]. It can be observed from Tables 1–4 that the  $\bar{h}^{\text{iso}}$  values obtained from the present correlation of  $Nu_d^{\text{iso}}$  shown in Eq. (25) agree with those obtained from Eq. (20). Of the 25 data used to develop this correlation, 22 fall about within  $\pm 6\%$  of the values obtained from Eq. (25). Table 4 also shows that the present estimates of the  $\bar{h}^{\text{iso}}$  values obtained from Eq. (25) agree with those given by Hu and Jacobi [6] for  $3320 \leq Re \leq 8440$  ( $5920 \leq Re_d \leq 15050$ ). This implies that the present inverse scheme should have good reliability and good accuracy for the present problem. Saboya and Sabora [22] have ever stated that the naphthalene vapor concentration at the subliming wall was very sensitive to the wall temperature, which was practically equal to the ambient temperature. In order to minimize experimental errors in the determination of the Sherwood numbers, the laboratory room was temperature controlled.

Once the average heat transfer coefficient on each sub-fin region is obtained, the temperature distribution on the fin can also be determined from Eq. (13). Because the average heat transfer coefficient on each sub-fin region is the approximate value, the temperature distribution on the fin is an approximate contour for various  $V_{\text{air}}$  and  $S$  values.

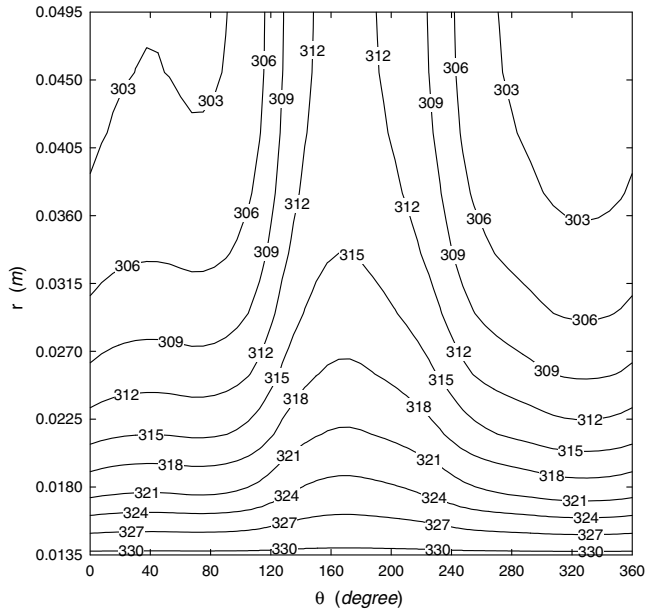


Fig. 6. Distribution of the calculated fin temperature for  $S = 0.005$  m and  $V_{air} = 1$  m/s.

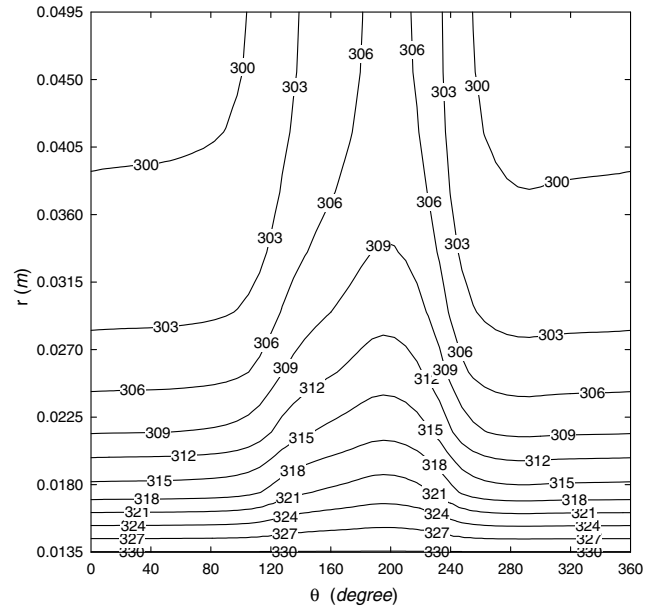


Fig. 8. Distribution of the calculated fin temperature for  $S = 0.005$  m and  $V_{air} = 5$  m/s.

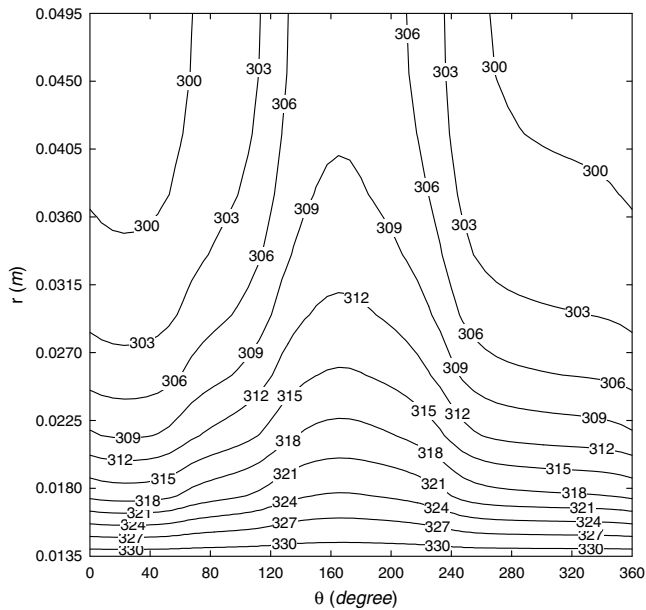


Fig. 7. Distribution of the calculated fin temperature for  $S = 0.005$  m and  $V_{air} = 3$  m/s.

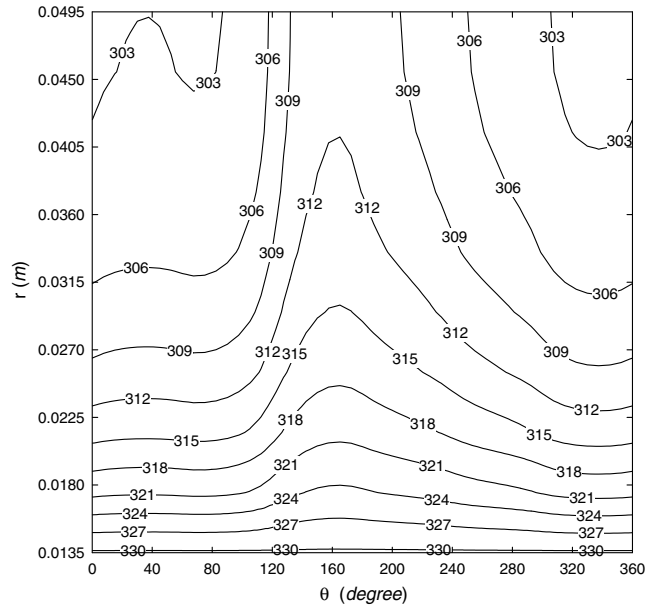


Fig. 9. Distribution of the calculated fin temperature for  $S = 0.015$  m and  $V_{air} = 1$  m/s.

Figs. 6–8, respectively, show the distributions of the calculated temperature on the fin for  $V_{air} = 1$  m/s, 3 m/s and 5 m/s as  $S = 0.005$  m. The distributions of the calculated fin temperature for  $V_{air} = 1$  m/s, 3 m/s and 5 m/s are, respectively, shown in Figs. 9–11 as  $S = 0.015$  m. It can be observed from Figs. 6–11 that, due to the poor thermal conductivity of the stainless fin, there is a considerable temperature drop between the tube wall and the edge of the fin especially on the upstream region. The fin temperature distributions obviously depart from the ideal isothermal situ-

ation and the fin temperature decreases more rapidly away from the circular center when  $V_{air}$  increases. Due to the above phenomenon, the fin efficiency decreases with increasing the frontal air speed or the Reynolds number.

6. Conclusions

The present study proposes a numerical inverse scheme involving the finite difference method in conjunction with the least-squares method and experimental fin tempera-

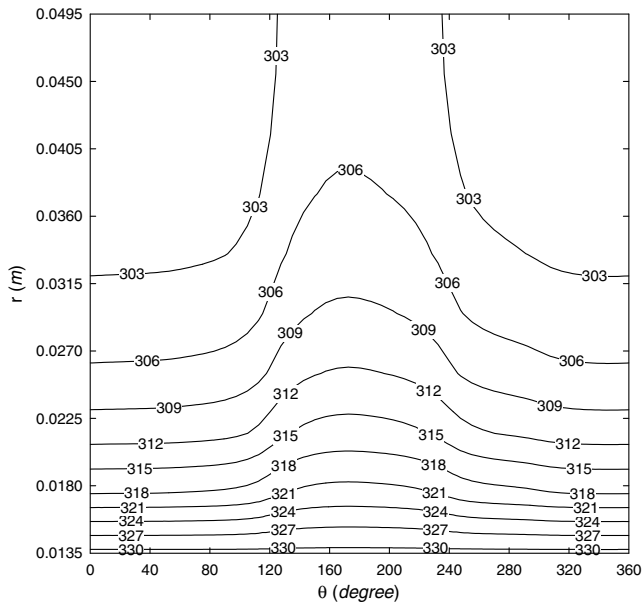


Fig. 10. Distribution of the calculated fin temperature for  $S = 0.015$  m and  $V_{\text{air}} = 3$  m/s.

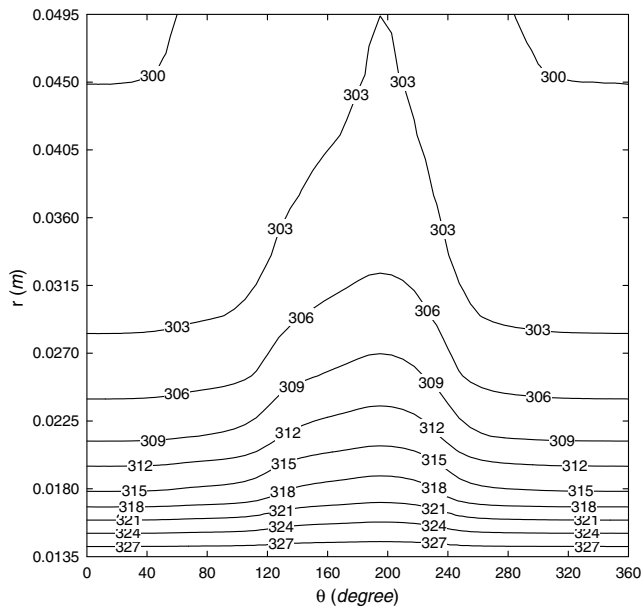


Fig. 11. Distribution of the calculated fin temperature for  $S = 0.015$  m and  $V_{\text{air}} = 5$  m/s.

tures to estimate the unknown heat transfer coefficients on sub-fin regions, average heat transfer coefficient on the whole fin  $\bar{h}$  and fin efficiency  $\eta_f$  for various  $V_{\text{air}}$  and  $S$  values. The estimated results show that there is a considerable temperature drop between the tube wall and the edge of the annular circular fin especially on the upstream region. The fin temperature distributions depart from the ideal isothermal situation and the fin temperature decreases more rapidly away from the circular center with increasing the air speed. The ratio of the average heat transfer coefficient on the front fin region to that on the wake fin region can be up to seven times for a given  $V_{\text{air}}$  value. The  $\bar{h}$  value

increases with increasing  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$  and increasing  $S$  for  $0.005 \text{ m} \leq S \leq 0.018 \text{ m}$ . However, the effect of the fin spacing  $S$  on the  $\bar{h}$  value can be negligible when the  $S$  value exceeds about  $0.018 \text{ m}$ . The  $\eta_f$  value decreases with increasing  $V_{\text{air}}$  for  $1 \text{ m/s} \leq V_{\text{air}} \leq 5 \text{ m/s}$  and seems to be not very sensitive to the fin spacing. An important finding is that the present estimated results can be applied to obtain the new correlations of the Nusselt number and fin efficiency based on  $d_o$ ,  $S$  and  $Re_d$ . The present estimates obtained from the correlation of the Nusselt number agree with those given by Hu and Jacobi [6] for  $3320 \leq Re \leq 8440$ . This implies that the present inverse scheme should have good reliability and good accuracy.

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